# IFF/SSR A.L.T.F.

# **Automatic Localised Test Facility**



ALTF enables a pre-flight test to be completed as the aircraft approaches the runway for take off. Details relating to the performance are made available to the pilot on a lamp matrix display system.

The test interrogator is a programmable low power system incorporating a solid state transmitter and logarithmic receiver. The transmission of interrogations is controlled by an interlaced mode encoder. Modes 1, 2, 3/A, 4 and C may be selected for check during each test sequence and will be priority assigned to suit operational needs. Replies from the transponder under test are detected by the logarithmic receiver and decoded in a synchronous decoder. Since the range of the aircraft under test is known, replies from that aircraft can be decoded and all others rejected.

Facilities on the test interrogator include a mimic of the pilot display and a 4 digit LED display for MKXA reply codes. The amplitude information in the reply train will assist in the evaluation of the aircraft antenna system. By reducing the test interrogator power output the performance of the transponder receiver can be established and by measuring the amplitude of the received reply from the transponder, the power output of that unit may be determined. The decoding provisions include All Mode facility and the provision of a Mode 4 interface to STANAG 4193 is made together with an internal space allocation for a Mode 4 memory module.

The antenna system is a three track broadside array designed to minimise radiation outside the test zone.

Aircraft antenna having heights from 2ft to 30ft above the ground are illuminated. The design offers a combination of low weight and size with the ability to withstand a high wind loading.

An equally important use for the Test Interrogator is for bench testing of transponder units. By connecting to the transponder, via a variable attenuator, a full unit functional test can be carried out. The results of the test can be down loaded via the RS232 interface.

# PA 6660 ampl will a anter

Full automatic testing covering all military and civil modes

Full system performance checks

> No aircraft modification necessary

> **Transponder** degradation

> > displayed

**Continual self test** 

facility

# **Electronic Systems Division**

# IFF/SSR A.L.T.F. PA 6660

# PA 6660 ALTF Installation

A full free space check is initiated measuring transponder parameters such as:

Transmitter power Receiver sensitivity Pulse widths Pulse spacings Frequency

Message content.

The ALTF system is designed to be installed 35m from the centreline of a standard 50ft wide taxiway and comprises:

# Display

A 300mm x 300mm matrix of 49 lenses which enables alpha-numeric information to be displayed for the pilot.

# **Antenna System**

Designed to equalise signals within the defined test area and to minimise radiation outside

# **Test Interrogator**

A specially designed transmitter/receiver unit with a microprocessor control system to define power levels, receiver sensitivities and also to control the signal processing.

# **Aircraft Sensor**

A small doppler radar having a low radiated power which detects an aircraft moving into the test area and initiates the test sequence

# **Data Summary**

### Mechanical Construction

# Dimensions

Test Interroga	ator	
Height: Width: Depth:	309mm 442mm 329mm (+10	)0mm cover)
Display Height: Width: Depth:	700mm 450mm 350mm	
Antenna Height:	1.6m	
Weight Test Interroga	ator	21.5Kg

21.5Kg
8.0Kg
21.0Kg
7.0Kg

# **Power Supplies**

AC Supplies	115/240V 50/60Hz Single phase
DC Supplies	22 to 32V
External Batteries	Sealed lead-acid

### **Temperature Range**

Operating -25°C to +55°C



# **Altitude Limits**

Operating Transportation

3000 metres 15000 metres

### **Test Interrogator** Transmitter

Frequency 1030 ± 0.2MHz Power Output Programmable over range -5 to -26dBm

### Receiver

Centre Frequency1090MHz Logarithmic Characteristic Dynamic Range 30dB

# Processing

Checks include:

Full mode capability i.e

M1, 2, 3/A, 4 and C

Full code capability 4096 codes

Transponder power output greater than 21dBw

Transponder sensitivity better than -69dBm

Transponder reply frequency 1090 + 3MHz

Transponder pulse interval Nominal  $\pm 0.15 \mu s$ 

Transponder SLS performance P2=P1+3dBP2=P1 -9dB

### Antenna System

Beamwidth Azimuth 60° Elevation 15°

# **Moving Aircraft Sensor**

Frequency 24.2GHz±10MHz Power output

@ Wavequide 5mW nominal

### Display

7x7 dot matrix providing:	
Fully operational transponder	
Confidence level 1 (high) Confidence level 2 Confidence level 3 Confidence level 4 (low)	White 10 White 9 White 8 White 7
Unserviceable Transponder	
Fail Transponder	Red FT
Fail Mode 1	Red F1
Fail Mode 2	Red F2
Fail Mode 3/A	Red F3
Fail Mode 4	Red F4
Fail Mode C	Red FC
Fail Antenna	Red A
ALTF unserviceable (BITE fail)	Red X
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Automatic night dimming incorporated

# **ALTF Installation**

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