

GEC-Marconi
AVIONICS

FLIGHT

I N T E R N A T I O N A L

BEHIND THE LINES

WITH

PHOENIX

**THE BRITISH ARMY'S
BATTLEFIELD SURVEILLANCE
AND
TARGET ACQUISITION SYSTEM**

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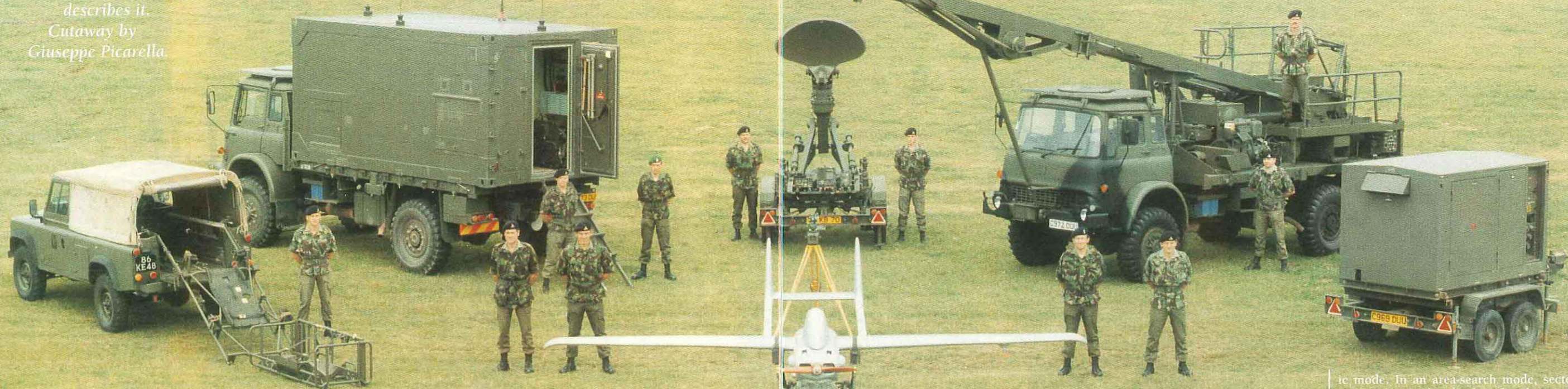
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BEHIND THE LINES

Almost a decade after contract award, and protracted development, GEC-Marconi Avionics' Phoenix unmanned air-vehicle is about to enter service. Simon Elliott describes it. Cutaway by Giuseppe Picarella.



Timely and accurate target intelligence is vital if maximum use is to be made of modern, long-range, artillery and multiple-launch-rocket systems. To provide this, commanders on the battlefield are coming increasingly to rely on the stealthy, low-key, unmanned air-vehicle (UAV).

The ideal UAV to support artillery would be able to provide surveillance in day and night conditions and in bad weather; it would have a real-time datalink; and would be operated easily by troops under battlefield conditions.

The Phoenix UAV, now in production for the Royal Artillery (RA), is such a surveillance system. The Phoenix was developed against a UK Ministry of Defence (MoD) staff target from a contract awarded in 1985. The MoD approved Phoenix design is being fit for RA acceptance in November 1993.

It has been designed from the outset as a user-friendly system for troops to operate in the field. George Haffey, Phoenix project manager at GMAV, says that a

principal aim was to make it simple to be operated and maintained by soldiers under battlefield conditions.

Initially designed for use with the RA's Multiple Launch Rocket System, the Phoenix is now likely to fulfil much wider surveillance application when it enters service in 1995.

The Phoenix will replace the current fleet of Canadair CL-89 Midge UAVs and Army maintenance instructors are attending training courses at GMAV before taking up their posts.

The Phoenix is an all-weather, day or night, real-time surveillance system which consists of a variety of elements. It was developed under a contract initially awarded in 1985. The twin-boom UAV provides surveillance through its mission pod, the imagery from which is datalinked via a ground data terminal (GDT) to a ground control station (GCS). This controls the overall Phoenix mission and is used to distribute the UAV provided intelligence direct to artillery forces, to command level, or to a Phoenix troop

command post (TCP). The principal method of communicating from the GCS to artillery is via the battlefield-artillery targeting engagement system (BATES).

Powered by an 19kW (25hp) Target Technology 342 two-stroke flat-twin engine, the Phoenix air-vehicle (with a centrally mounted fuel tank) is almost entirely manufactured from composites such as Kevlar, glassfibre and carbonfibre reinforced plastic and Nomex honeycomb. The principal sub-contractor on the airframe is Flight Refuelling of Christchurch, Dorset, UK.

The Phoenix air vehicle is of modular design, for ease of transportation on the battlefield, being broken down into the fuselage with the twin tail-booms; detachable wings; tailplane attachment; a dorsal shock absorber (for landing); the mission pod; and replaceable wing and fin tips. The UAV can be launched within 1h upon reaching a launch site and a second can be despatched 8min later, from the same launcher. The wing span is 5.5m and max-

imum launching weight 175kg. GEC says: "Flight endurance is in excess of 4h, radius [of action] more than 50km and maximum altitude 9,000ft [2,750m]." The maximum speed is 85kts (155km/h).

Before launch and recovery, the UAV is stored in an electromagnetic-pulse (EMP)-hardened container, each vehicle having a shelf life of 15 years.

The Phoenix UAV system is sensor-oriented, the air vehicle operating as a taxi to carry the sensor payload to and around the designated target area. The 50kg mission pod, slung beneath the fuselage, features a two-axis stabilised turret carrying a thermal-imaging sensor. The latter is based on the GMAV Thermal Imaging Common Modules II hardware and features a 60° x 40° field-of-view Sprite detector operating in the 8-14µm band. Using a zoom telescope, magnification is provided between x2.5 and x10. The sensor is designed to scan through 360° continuously in azimuth and elevation.

During the cruise phase of the mission, the sensor can be locked at a pre-set elevation or prepared to scan in an automat-

ic mode. In an area-search mode, sector scanning is used and, when the Phoenix is in orbit round a target, the sightline is steered automatically to ensure that it remains correctly targeted.

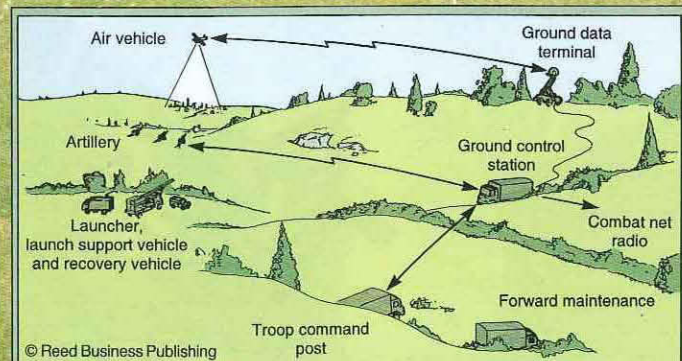
The mission pod also has the airborne components of the datalink. These are automatically switchable steered antennae fore and aft, "giving highly directional secure communications", says GMAV.

LEVEL HORIZON

A third roll axis is provided to stabilise the whole sensor pod so that both the turret-mounted sensor and the datalink antennas are always horizontal. This gives the ground-based observer a level horizon and simplifies control of the datalink. "The operator on the ground has no perception of the UAV turning and so suffers no motion disorientation," says Haffey.

The Phoenix sensor-pod package is also flexible. GMAV says: "Interchangeable, alternative, pods could provide early warning, air-defence suppression, laser designation, communication relay, decoy or nuclear, biological or chemical [NBC] monitoring roles."

The level of command above the GCS



Battlefield surveillance scenario

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Phoenix UAV

- 1 Dog clutch type starter
- 2 Two blade fixed pitch wooden propeller(780mm)
- 3 Four point anti-vibration engine mount
- 4 WAE 342, 342cc Flat twin, fuel injection two-stroke engine
- 5 Exhaust outlet
- 6 Fuel regulator
- 7 Fuel pump
- 8 Throttle actuator
- 9 Engine Management Unit (EMU)
- 10 Engine bulkhead
- 11 Electrical connection (EMU)
- 12 Electrical connection for engine driven 900w Plessey generator
- 13 Hinges
- 14 Fuel supply line
- 15 Fuel return line
- 16 Outside air temperature probe
- 17 Parachute ground stroop release mechanism
- 18 Engine bulkhead/fuselage catch
- 19 Flight control computer
- 20 Connector for ground impact detection circuit
- 21 Battery pack
- 22 Rate gyro pack
- 23 Airspeed transducer
- 24 Altitude transducer
- 25 Vertical gyro
- 26 Power distribution unit
- 27 Pod roll actuator
- 28 Magnetometer electronic unit
- 29 Fuel stop valve
- 30 Parachute forward suspension arm
- 31 Pod forward pick-up point
- 32 Removable, limited shock-absorbing wingtip
- 33 Wingtip latch receptical
- 34 Wingtip locating spigots
- 35 Wingtip connection hooks
- 36 Aileron housing and spigot hinge
- 37 Aileron actuator
- 38 Aileron actuator arm
- 39 Frangible shock absorber module with foam insert
- 40 Ground impact detector circuit
- 41 Electrical connector for 40
- 42 Shock absorber locating spigots
- 43 Fuel cap cut-out
- 44 Wing forward spigot and locating hook
- 45 Wing rear spigot and locating hook
- 46 Fuel manifold assembly
- 47 Fuel-level indicator (variable-resistance float type)
- 48 Fuel tank
- 49 Pod locking mechanism
- 50 Pod locking arm
- 51 Pod/taxi electrical connector
- 52 Pitot tube with static vents
- 53 Wingtip spigot housing
- 54 Wingtip locating hinges
- 55 Removable port-wingtip
- 56 Aileron spigot hinge
- 57 Wingtip latch assembly

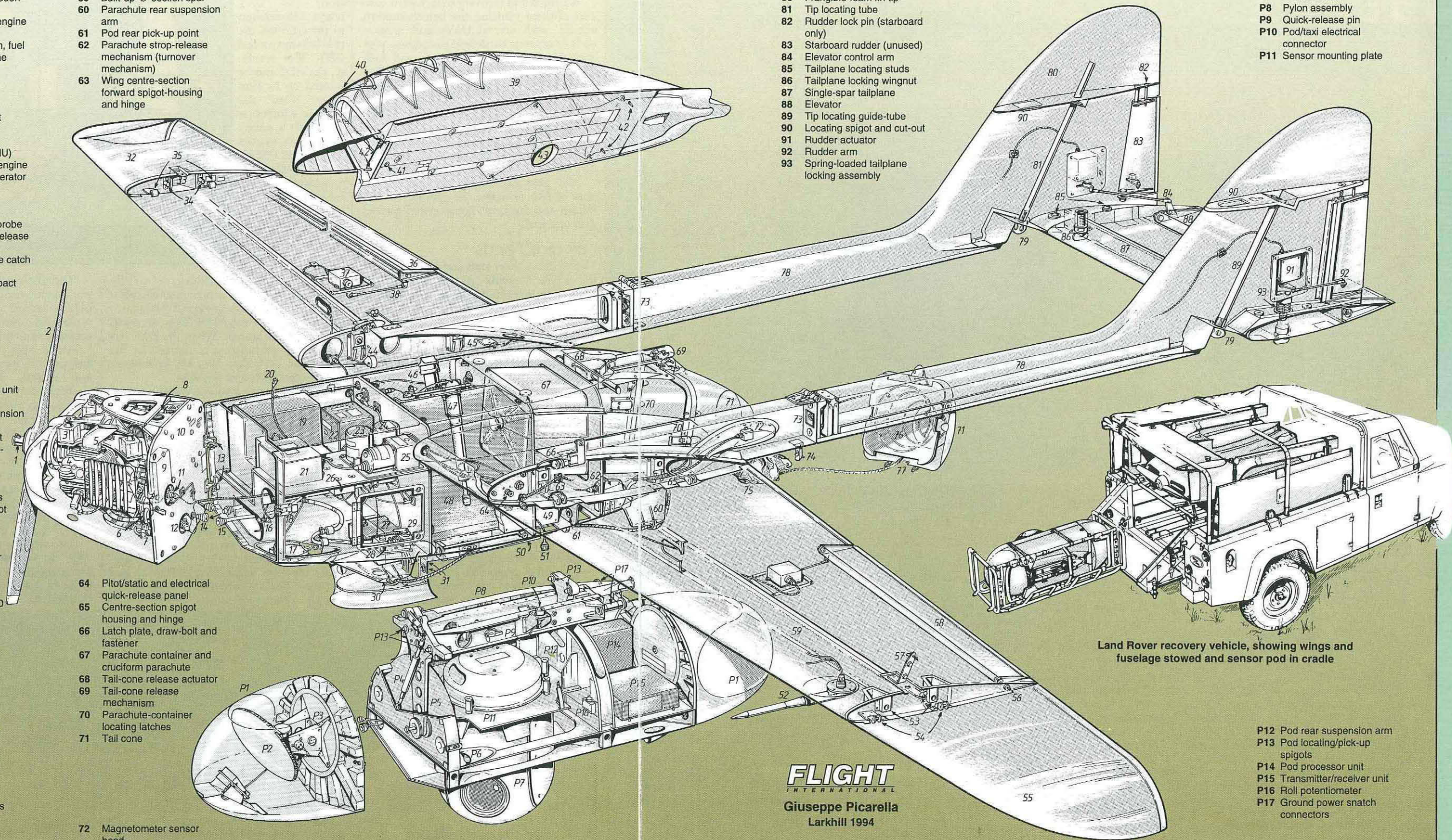
- 58 Port aileron
- 59 Built-up 'C' section spar
- 60 Parachute rear suspension arm
- 61 Pod rear pick-up point
- 62 Parachute stroop-release mechanism (turnover mechanism)
- 63 Wing centre-section forward spigot-housing and hinge

- 64 Pitot/static and electrical quick-release panel
- 65 Centre-section spigot housing and hinge
- 66 Latch plate, draw-bolt and fastener
- 67 Parachute container and cruciform parachute
- 68 Tail-cone release actuator
- 69 Tail-cone release mechanism
- 70 Parachute-container locating latches
- 71 Tail cone

- 72 Magnetometer sensor head
- 73 Fuselage/tailboom hinge and electrical connections

- 74 Fuselage/tailboom locking latch
- 75 Drone parachute
- 76 Tail-cone spring-loaded ejection plate
- 77 Tail-cone locating arms
- 78 Interchangeable tailbooms
- 79 Fin-tip retaining wing nut
- 80 Frangible-foam fin tip
- 81 Tip locating tube
- 82 Rudder lock pin (starboard only)
- 83 Starboard rudder (unused)
- 84 Elevator control arm
- 85 Tailplane locating studs
- 86 Tailplane locking wingnut
- 87 Single-spar tailplane
- 88 Elevator
- 89 Tip locating guide-tube
- 90 Locating spigot and cut-out
- 91 Rudder actuator
- 92 Rudder arm
- 93 Spring-loaded tailplane locking assembly

- P1 Antenna housing
- P2 Antenna plate
- P3 Antenna plate assembly and servos
- P4 Pod roll arm
- P5 Pod forward suspension arm
- P6 Electrical cooling air inlet
- P7 Sensor turret
- P8 Pylon assembly
- P9 Quick-release pin
- P10 Pod/taxi electrical connector
- P11 Sensor mounting plate



Land Rover recovery vehicle, showing wings and fuselage stowed and sensor pod in cradle

FLIGHT
INTERNATIONAL
Giuseppe Picarella
Larkhill 1994

- P12 Pod rear suspension arm
- P13 Pod locating/pick-up spigots
- P14 Pod processor unit
- P15 Transmitter/receiver unit
- P16 Roll potentiometer
- P17 Ground power snatch connectors

PHOENIX DESCRIBED

is the TCP, which controls up to three flight sections. Each flight section consists of a launch-and-recovery detachment and a ground-control detachment.

A launch and recovery detachment consists of three vehicles; the launch-support vehicle, with several UAVs and mission pods in separate battlefield containers, plus operational replacement spares and fuel; the launch vehicle, which features a pallet-mounted lifting crane, the hydraulic/pneumatic catapult and launch ramp, a pre-launch air-vehicle data-entry device, built-in test equipment and air-vehicle warm-up unit and engine starter; and the Land Rover recovery vehicle, which is fitted with cradles for the air vehicle and mission pod.

The ground-control detachment consists of two vehicles, the ground control station and the Land Rover-towed ground data terminal.

CONTROL FOCUS

The GCS has three staff and is the focus of the whole Phoenix system. Able to be located up to 25km from the UAV launcher, it provides command, control and communications from an NBC- and EMP-hardened truck-mounted shelter. The GCS features three "...identical, state-of-the-art networked workstations for the mission controller, image analyst and air-vehicle controller. Their flicker-free, high-resolution VDUs [visual-display units] are menu-driven to simplify operation and minimise operator workload", says GMAV.

Each screen uses a common format, with a menu on the right-hand side which allows either the datalinked thermal-imaging picture from the UAV or a multi-scale map showing target and UAV positions to be viewed.

Phoenix can be flown either autonomously or by command from the operator, who requires no piloting skills. The sensor is controlled directly from the GCS, by means of one of several automatic pointing modes, to aid target detection,

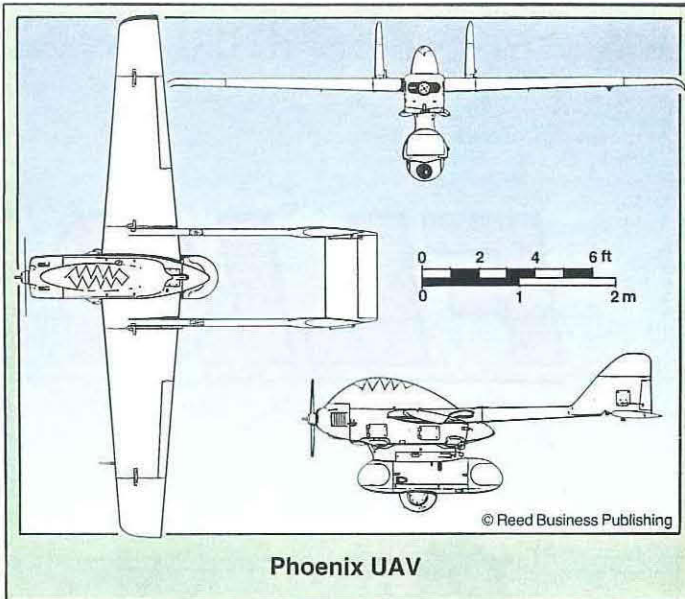
recognition and identification.

The J-band secure datalink from the UAV relays information to the GCS via the trailer-mounted GDT. "For security, the GDT can be located up to 1km away from the GCS, to which it is linked by a secure communications cable," says GMAV.

Haffey describes a typical Phoenix mission: "Once the UAV has been assembled and placed on the launcher, information from the GCS is entered, which enables it to fly autonomously on to a pre-assigned radial after launch. It is also given instructions on an emergency recovery location in the event of failure after launch. When the UAV has flown on to its radial, it is acquired by the GDT. The UAV then flies via its mission plan to the target-area artillery."

Haffey says of a typical engagement scenario: "The grid co-ordinates of a target located by the UAV are fed through the GCS and BATES directly to the guns. The UAV notes the corresponding fall of shot, corrections are automatically generated at the GCS, fed via BATES to the guns and the target re-engaged."

When the mission is complete, recovery is by parachute, with the UAV inverting before impact and landing on its back to protect the mission pod. The actual impact force is absorbed by a crushable dorsal shock absorber and by frangible wing and tail tips. The shock absorber and tips are all quickly replaceable.



Phoenix UAV

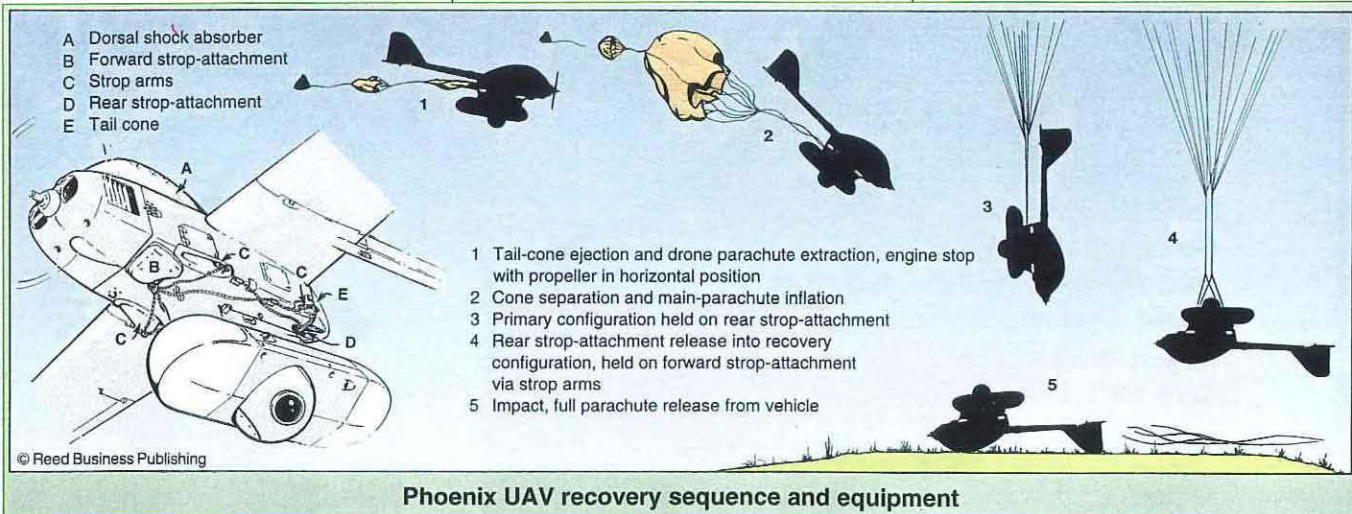
The whole system is supported by a forward maintenance unit, which, in a battlefield-hardened shelter, contains line-replaceable-unit test equipment and accommodation for repairs.

GMAV also believes that the Phoenix has excellent export potential, with the system already being offered for the Netherlands requirement for a UAV. Haffey admits that alterations may be needed to suit some export customers, however.

He says: "For hot-and-high performance, the powerplant needs to be uprated." GMAV is examining alternative engines, probably of rotary design.

With the export potential and what Haffey describes as a "decent production run", GMAV believes that it has a winner in the Phoenix UAV system.

The likelihood is that, by the end of 1995, it will have seen service with UK forces deployed overseas — which is the ultimate test for the UK's most potent battlefield UAV. ✻



Phoenix UAV recovery sequence and equipment